Dynamic Behaviour Analysis of a Dual-Rotor System Using the Transfer Matrix Method

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This paper presents a general formulation for the problem of the steady-state unbalance response of a dual rotor system with a flexible intershaft bearing using an 'extended' transfer matrix method, where the transfer matrix assumes a dimension of (33×33) . The validity of the formulation is established by comparing the results obtained through a computer program with closed form solutions available for some simple cases. Some interesting phenomena of steady-state whirl orbits of the dual rotor system are described.

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Nomenclature

[A], [I]	3] Represent transfer matrices,
[C], [L]	$D] \int each of dimension (33 \times 33)$
$[\bar{B}]$	– Bearing matrix
C_{yy}, C_{zz}	 Direct damping coefficients
C_{yz}, C_{zy}	- Cross coupled damping coefficients
C_H, C_L	- High and low rotor damping, respectively
Ε	 Modulus of elasticity
е	- Eccentricity
[F]	– Field matrix
$[\bar{F}]$	– Overall field matrix
h	 Thickness of the disk
[<i>I</i>]	– Unit matrix
I_T	 Transverse mass moment of inertia of the disk
I_P	 Polar mass moment of inertia of the disk
K_{yy}, K_{zz}	 Direct stiffnesses
K_{yz}, K_{zy}	 Cross coupled stiffnesses
K_H, K_L	- High and low rotor stiffness, respectively
1	 Sectional length
M_y, M_z	– Moments about <i>y</i> - and <i>z</i> -axes, respectively
M_H, M_L	- High and low rotor mass, respectively
т	 Mass of the element per unit length
m_b	 Mass of the shaft at bearing station
m_e	– Unbalance mass
$[\underline{P}]$	– Point matrix
[P]	 Overall point matrix
U_y, U_z	 Unbalance components
V_y, V_z	- Shear force in <i>y</i> - and <i>z</i> -directions, respectively
<i>v</i> , <i>w</i>	– Deflection along <i>y</i> - and <i>z</i> -axes, respectively
X, Y, Z	 Fixed co-ordinate system
$\bar{X}, \bar{Y}, \bar{Z}$	 Rotating co-ordinate system

 $\left\{ S \right\}_{OA}, \left\{ S \right\}_{OB} \\ \left\{ S \right\}_{OC}, \left\{ Z \right\}_{OD} \right\} -$ Represent end conditions of the dual rotor system

 ω_m, ω_n – Inner and outer rotor speeds, respectively

- Slope in x-z and y-z planes, respectively θ, ϕ

- Angular position of unbalance β

λ - Eigenvalue

Subscripts

m, n	 Pertaining to the effect of inner and outer rotor speeds, respectively 	
с	– Cosine component	
ст, с	 cn – Cosine components of inner and outer rotor, respectively 	
S	– Sine component	
sm, s	 Sine components of inner and outer rotor, respectively 	
у	– Along <i>y</i> -axis	
Ζ	– Along <i>z</i> -axis	
j	 Pertaining to a junction 	
Superscripts		
R	– Right to a section	
L	– Left to a section	
t	– Transpose of a matrix	
1. INTRODUCTION		
V	With the ever increasing demand of large power and	

smaller gas turbine engines for aircraft propulsion, a two spool system with intershaft bearings is becoming a standard layout to accommodate the compressor and turbine rotors.